



Fluid Mechanics CEE 3311

LECTURE 10

Conservation of energy

**Simplified forms of the  
energy equation**

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## Simplified forms of the energy equation

### Energy equation for steady, one dimensional incompressible flow in a pipe

Consider flow through the pipe system shown below.

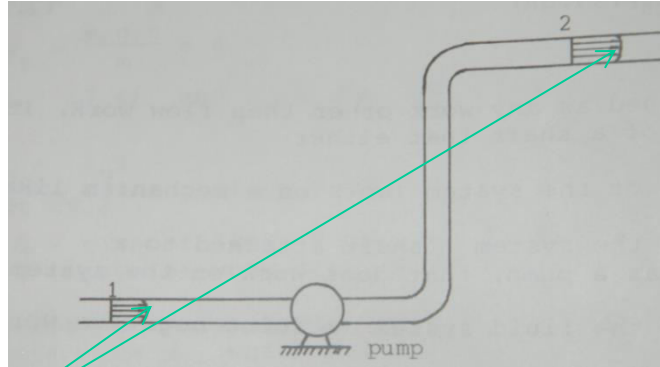


Fig 11.1 Pipe system

$$\frac{dQ}{dt} - \frac{dW_s}{dt} = \frac{d}{dt} \int_{cv} \left( \frac{v^2}{2} + gz + u \right) \rho dV + \int_{cs} \left( \frac{v^2}{2} + gz + u + \frac{p}{\rho} \right) \rho (\vec{v} \cdot d\vec{A}) \quad 11.1$$

For a steady flow situation in which there is one entrance and one exit across which **uniform** profiles can be assumed

$$\frac{dQ}{dt} - \frac{dW_s}{dt} = 0 + \int_{cs} \left( \frac{v^2}{2} + gz + u + \frac{p}{\rho} \right) \rho (\vec{v} \cdot d\vec{A}) \quad 11.2$$

$$\frac{dQ}{dt} - \frac{dW_s}{dt} = \int_{A_1} \left( \frac{v_1^2}{2} + gz_1 + u_1 + \frac{p_1}{\rho_1} \right) \rho_1 (-v_1 dA_1) + \int_{A_2} \left( \frac{v_2^2}{2} + gz_2 + u_2 + \frac{p_2}{\rho_2} \right) \rho_2 (v_2 dA_2) \quad 11.3$$

$$\frac{dQ}{dt} - \frac{dW_s}{dt} = \int_{A_1} \left( \frac{v_1^2}{2} + gz_1 + u_1 + \frac{p_1}{\rho_1} \right) \rho_1 (-v_1 dA_1) + \int_{A_2} \left( \frac{v_2^2}{2} + gz_2 + u_2 + \frac{p_2}{\rho_2} \right) \rho_2 (v_2 dA_2) \quad 11.3$$

For a flow the term  $\left( \frac{v^2}{2} + gz + \frac{p}{\rho} \right)$  in eq is constant across the cross section because  $v$  is constant (we assume a uniform velocity profile) and the sum of  $\frac{p}{\rho} + gz$  is constant if the streamlines at each section are parallel. Therefore, we take the term outside the integral and separating the velocity term integral

$$\frac{dQ}{dt} - \frac{dW_s}{dt} + \left( \frac{p_1}{\rho_1} + gz_1 + u_1 \right) \int_{A_1} \rho_1 v_1 dA_1 + \int_{A_1} \frac{\rho_1 v_1^3}{2} dA_1 = \left( \frac{p_2}{\rho_2} + gz_2 + u_2 \right) \int_{A_2} \rho_2 v_2 dA_2 + \int_{A_2} \frac{\rho_2 v_2^3}{2} dA_2 \quad 11.4$$

It can be seen that  $\int \rho v dA = \rho \bar{v} A = \frac{dm}{dt}$  = mass rate of flow. Where  $\bar{v}$  is the mean or average velocity over a cross section.

However,  $\frac{dm}{dt}$  does not appear as a factor of  $\int \frac{\rho v^3}{2} dA$  in

eq 11.4; so it is common to express  $\int \frac{\rho v^3}{2} dA$  as

$\alpha \rho \left( \frac{\bar{v}^3}{2} \right) A$ . equation 11.4 now becomes

$$\alpha \rho \frac{\bar{v}^3}{2} A = \alpha \frac{\bar{v}^{-2}}{2} \rho \bar{v} A = \alpha \frac{\bar{v}^{-2}}{2} \frac{dm}{dt}$$

$$\frac{dQ}{dt} - \frac{dW_s}{dt} + \left( \frac{p_1}{\rho_1} + gz_1 + u_1 \right) \frac{dm}{dt} + \alpha_1 \frac{\bar{v}_1^2}{2} \frac{dm}{dt} = \left( \frac{p_2}{\rho_2} + gz_2 + u_2 \right) \frac{dm}{dt} + \alpha_2 \frac{\bar{v}_2^2}{2} \frac{dm}{dt} \quad 11.5$$

See also Lecture 8

$\alpha = \frac{1}{A} \int \left( \frac{v}{\bar{v}} \right)^3 dA =$   
kinetic energy correction factor  
 $\alpha = 1$  for uniform flow ( $v = \bar{v}$ )  
For most cases of turbulent flow,  $\alpha \approx 1.05$ . because this is very close to unity, it is common practice in engineering applications to let  $\alpha = 1$ .

$$\frac{dQ}{dt} - \frac{dW_s}{dt} + \left( \frac{p_1}{\rho_1} + gz_1 + u_1 \right) \frac{dm}{dt} + \alpha_1 \frac{\bar{v}_1^2}{2} \frac{dm}{dt} = \left( \frac{p_2}{\rho_2} + gz_2 + u_2 \right) \frac{dm}{dt} + \alpha_2 \frac{\bar{v}_2^2}{2} \frac{dm}{dt}$$

$$\frac{dQ}{dt} - \frac{dW_s}{dt} + \left( \frac{p_1}{\rho_1} + gz_1 + u_1 + \alpha_1 \frac{\bar{v}_1^2}{2} \right) \frac{dm}{dt} = \left( \frac{p_2}{\rho_2} + gz_2 + u_2 + \alpha_2 \frac{\bar{v}_2^2}{2} \right) \frac{dm}{dt} \quad 11.6$$

Dividing by  $\frac{dm}{dt}$

$$\frac{1}{\frac{dm}{dt}} \left( \frac{dQ}{dt} - \frac{dW_s}{dt} \right) + \frac{p_1}{\rho_1} + gz_1 + u_1 + \alpha_1 \frac{\bar{v}_1^2}{2} = \frac{p_2}{\rho_2} + gz_2 + u_2 + \alpha_2 \frac{\bar{v}_2^2}{2} \quad 11.7$$

The shaft work term is usually the result of a turbine or pump in the flow system. It is therefore convenient to represent the shaft work term as

$$W_s = W_t - W_p \quad 11.8$$

where  $W_t$  = power delivered by a turbine

$W_p$  = power supplied by a pump

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In the latter case, the fluid system is doing negative work on its surrounding.

Substituting eq 11.8 into eq 11.7 and divided by g results in

$$\frac{\frac{dW_p}{dt}}{g \frac{dm}{dt}} + \frac{p_1}{\rho g} + z_1 + \alpha_1 \frac{v_1^2}{2g} = \frac{\frac{dW_t}{dt}}{g \frac{dm}{dt}} + \frac{p_2}{\rho g} + z_2 + \alpha_2 \frac{v_2^2}{2g} + \left\{ \frac{1}{g} (u_2 - u_1) - \frac{\frac{dQ}{dt}}{g \frac{dm}{dt}} \right\} \quad 11.9$$

$$\frac{dW_p}{g \frac{dm}{dt}} + \frac{p_1}{\rho g} + z_1 + \alpha_1 \frac{v_1^2}{2g} = \frac{dW_t}{g \frac{dm}{dt}} + \frac{p_2}{\rho g} + z_2 + \alpha_2 \frac{v_2^2}{2g} + \left\{ \frac{1}{g} (u_2 - u_1) - \frac{dQ}{g \frac{dm}{dt}} \right\} \quad 11.9$$

Replacing  $g(dm/dt)$  by  $dw/dt$

$$\frac{dW_p}{\frac{dw}{dt}} + \frac{p_1}{\rho g} + z_1 + \alpha_1 \frac{v_1^2}{2g} = \frac{dW_t}{\frac{dw}{dt}} + \frac{p_2}{\rho g} + z_2 + \alpha_2 \frac{v_2^2}{2g} + \left\{ \frac{1}{g} (u_2 - u_1) - \frac{dQ}{\frac{dw}{dt}} \right\} \quad 11.10$$

Where  $\frac{dw}{dt} = g \frac{dm}{dt}$  weight flow rate (from  $w = gm$ )

- $$\frac{\frac{dW_p}{dt}}{\frac{dw}{dt}} + \frac{p_1}{\rho g} + z_1 + \alpha_1 \frac{v_1^2}{2g} = \frac{\frac{dW_t}{dt}}{\frac{dw}{dt}} + \frac{p_2}{\rho g} + z_2 + \alpha_2 \frac{v_2^2}{2g} + \left\{ \frac{1}{g} (u_2 - u_1) - \frac{\frac{dQ}{dt}}{\frac{dw}{dt}} \right\} \quad 11.10$$

- All of the terms of eq 11.10 have one dimension:

- $$\frac{\text{energy}}{\text{unit weight of fluid}} = \frac{[\text{FORCE}][\text{LENGTH}]}{[\text{FORCE}]} = [\text{LENGTH}]$$

Hence  $\frac{\frac{dW_p}{dt}}{\frac{dw}{dt}}$  may be designated as  $h_p$  ( **pump head** ) and similarly  $\frac{\frac{dW_t}{dt}}{\frac{dw}{dt}}$  as  $h_t$  ( **turbine head** ).

The term  $\left\{ \frac{1}{g} (u_2 - u_1) - \frac{\frac{dQ}{dt}}{\frac{dw}{dt}} \right\}$  represents a loss of mechanical energy due to viscous stresses, which is usually lumped together in a single term called head loss and symbolised by  $h_l$ . Thus eq 11.10 becomes

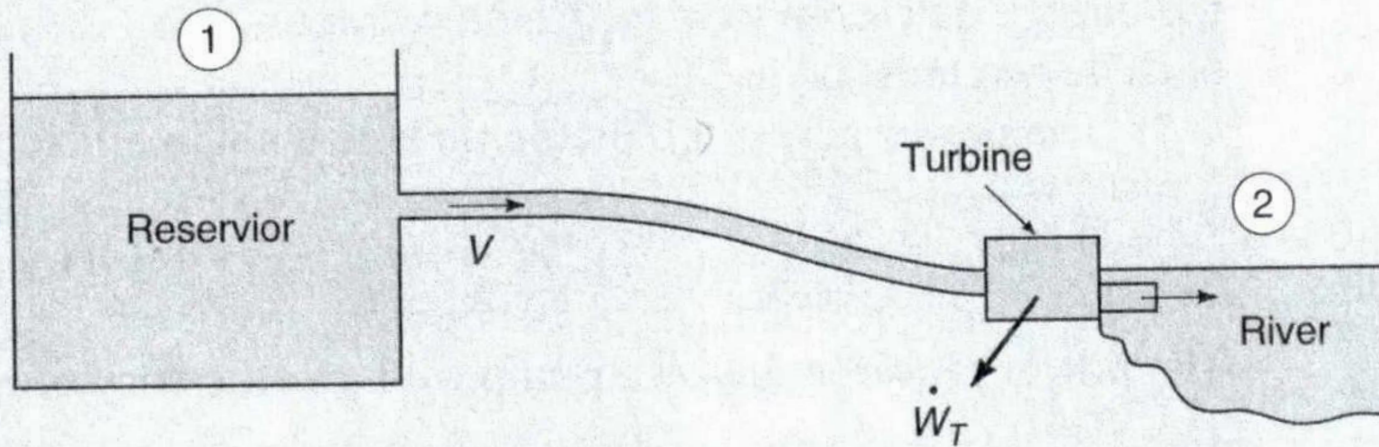
$$\frac{p_1}{\rho g} + z_1 + \alpha_1 \frac{v_1^2}{2g} + h_p = \frac{p_2}{\rho g} + z_2 + \alpha_2 \frac{v_2^2}{2g} + h_t + h_l \quad 11.11$$

This is the steady flow energy equation.

$$\alpha_1 \frac{v_1^2}{2g} + \frac{p_1}{\rho g} + z_1 + h_p = \alpha_2 \frac{v_2^2}{2g} + \frac{p_2}{\rho g} + z_2 + h_t + h_l$$

- The sum of the terms on the left hand side of eq represents the total energy, stated in energy per unit weight of flowing liquid, plus the *energy supplied* by a pump.
- The sum of the terms on the right hand side represents the total energy per unit weight at the downstream section plus the *energy given up* to a turbine and energy lost to friction between the two sections

Water flows from a reservoir through a 0.8-m-diameter pipeline to a turbine-generator unit and exits to a river that is 30 m below the reservoir surface (Fig. E4.7). If the flow rate is



**Figure E4.7**

3 m<sup>3</sup>/s and the turbine-generator efficiency is 80%, calculate the power output. Assume the loss coefficient in the pipeline (including the exit) to be  $K = 2$ .

**Solution:** The control volume to be used extends from section 1 to section 2; we consider the water surface of the left reservoir to be the entrance and the water surface of the river to be the exit. Because we assume the water surfaces to be large, the velocities at the surfaces are negligible. The velocity in the pipe is

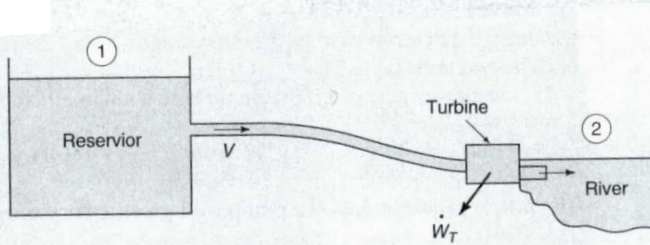


Figure E4.7

$$V = \frac{Q}{A}$$

$$= \frac{3}{\pi \times 0.8^2/4} = 5.968 \text{ m/s}$$

Now, consider the energy equation. We will use gage pressures so that  $p_1 = p_2 = 0$ ; the datum is placed through the lower section 2 so that  $z_2 = 0$ ; the velocities  $V_1$  and  $V_2$  are negligibly small;  $K$  is assumed to be based on the 0.8-m-diameter pipe velocity. The energy equation (4.4.24) then becomes

$$\cancel{H_P} + \cancel{\frac{V_1^2}{2g}} + \cancel{\frac{p_1}{\gamma}} + z_1 = H_T + \cancel{\frac{V_2^2}{2g}} + \cancel{\frac{p_2}{\gamma}} + \cancel{z_2} + K \frac{V^2}{2g}$$

$$30 = H_T + 2 \frac{5.968^2}{2 \times 9.81}$$

$$\therefore H_T = 26.4 \text{ m}$$

From this the power output is found, using Eq. 4.4.25, to be

$$\begin{aligned}\dot{W}_T &= Q\gamma H_T\eta_T \\ &= 3 \times 9810 \times 26.4 \times 0.8 = 622\,000 \text{ W} \quad \text{or} \quad 622 \text{ kW}\end{aligned}$$

In this example we have used gage pressure; the potential-energy datum was assumed to be placed through section 2,  $V_1$  and  $V_2$  were assumed to be insignificantly small, and  $K$  was assumed to be based on the 0.8-m-diameter pipe velocity.

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# Energy equation for non-viscous steady, one dimensional incompressible flow in a pipe

□ If :

- the losses are negligible
- there is no shaft work
- the flow is incompressible

The energy equation becomes  $\alpha_1 \frac{v_1^2}{2g} + \frac{p_1}{\rho g} + z_1 + h_p = \alpha_2 \frac{v_2^2}{2g} + \frac{p_2}{\rho g} + z_2 + h_t + h_l$

$$\frac{v_1^2}{2g} + \frac{p_1}{\rho g} + z_1 = \frac{v_2^2}{2g} + \frac{p_2}{\rho g} + z_2$$

11.12

The energy equation has been reduced to a form identical with the Bernoulli equation. Assumptions are similar in both equations.